

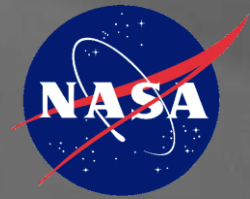
# Frequency Division Multiplexed Strain Sensor System

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NASA Langley Research Center

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Washington, DC



# Outline

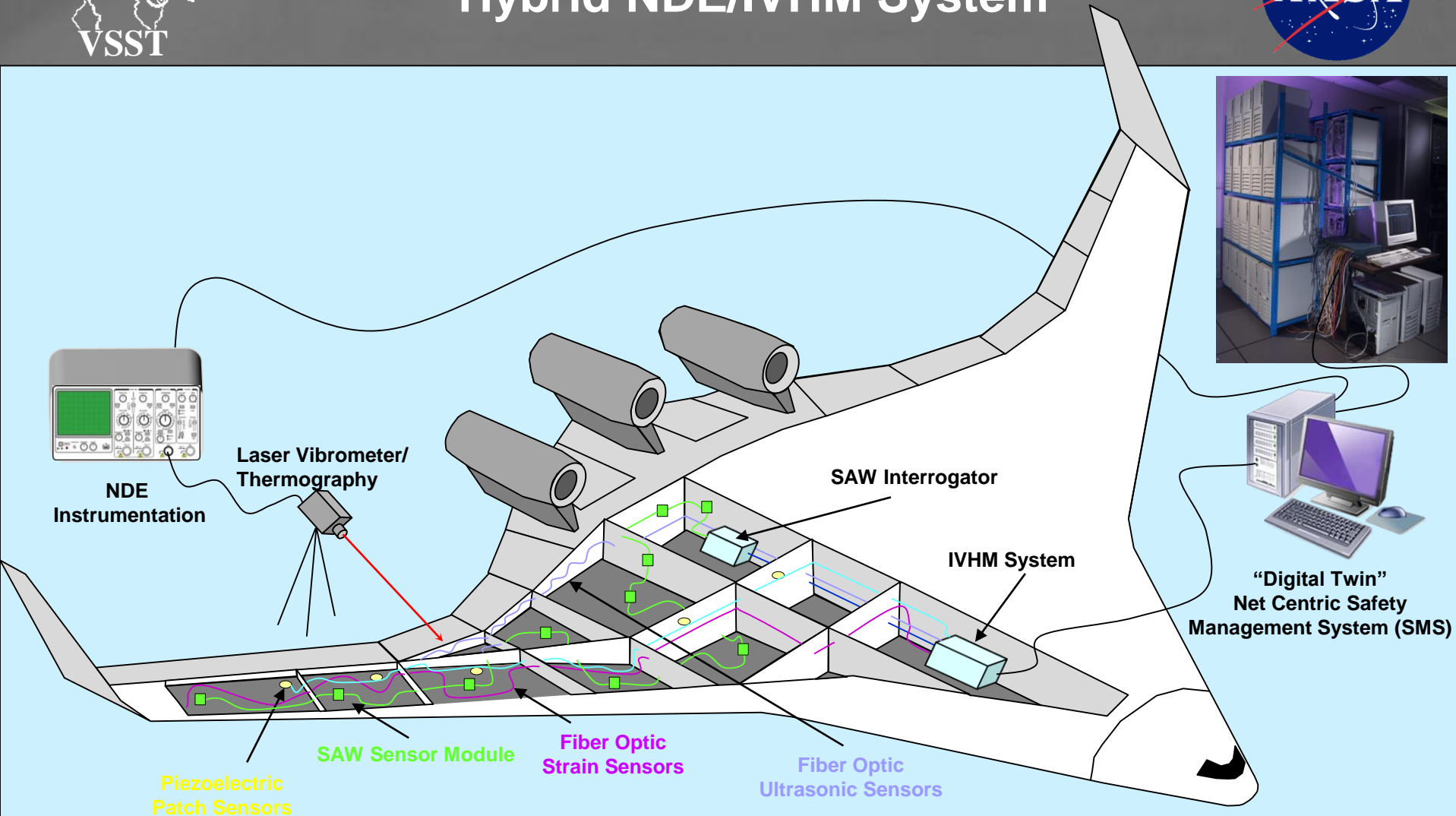
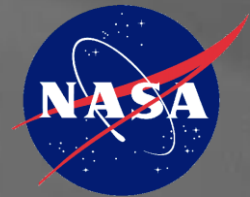


- **Project Goals**
- **FDM SAW Modules**
- **SAW Sensor Details**
- **FDM Strain Sensor System**
- **Results**
- **Conclusions**
- **Funding/Partnership Opportunities**



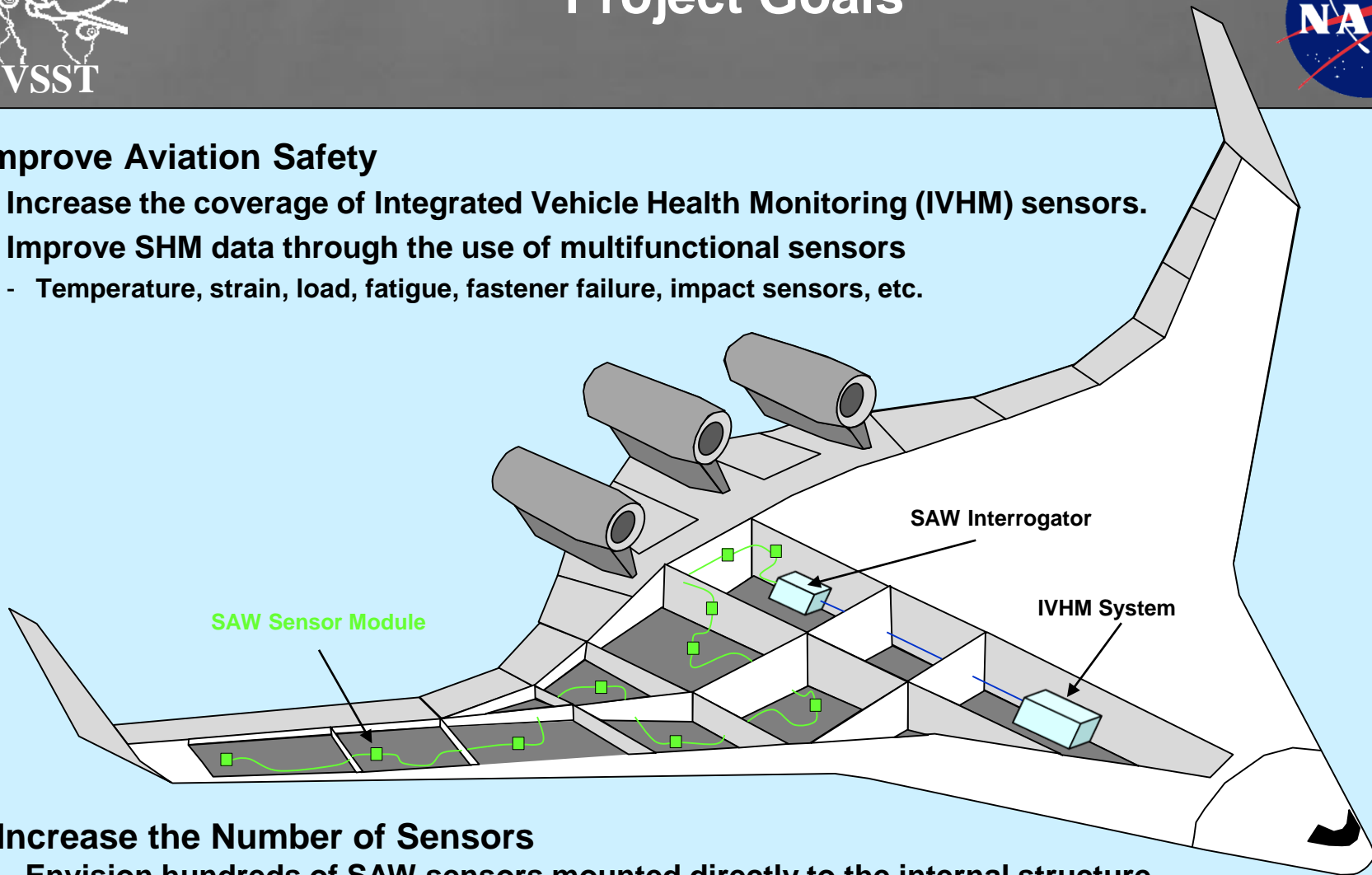


# The Big Picture Hybrid NDE/IVHM System



**Hybrid NDE/IVHM - Combines traditional ground NonDestructive Evaluation (NDE) methods with Integrated Vehicle Health Management Systems (IVHM) through the use of external loads and/or excitation methods combined with fixed sensors and onboard IVHM systems.**

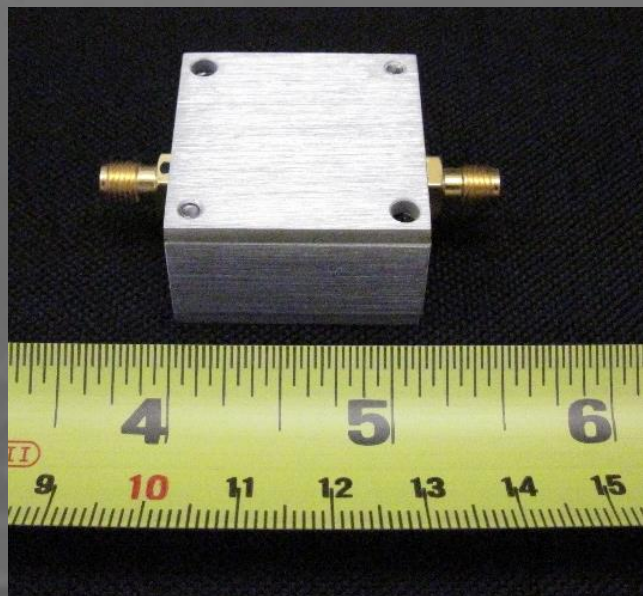
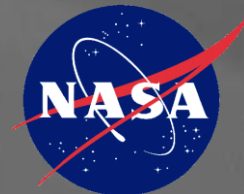
- **Improve Aviation Safety**
  - Increase the coverage of Integrated Vehicle Health Monitoring (IVHM) sensors.
  - Improve SHM data through the use of multifunctional sensors
    - Temperature, strain, load, fatigue, fastener failure, impact sensors, etc.



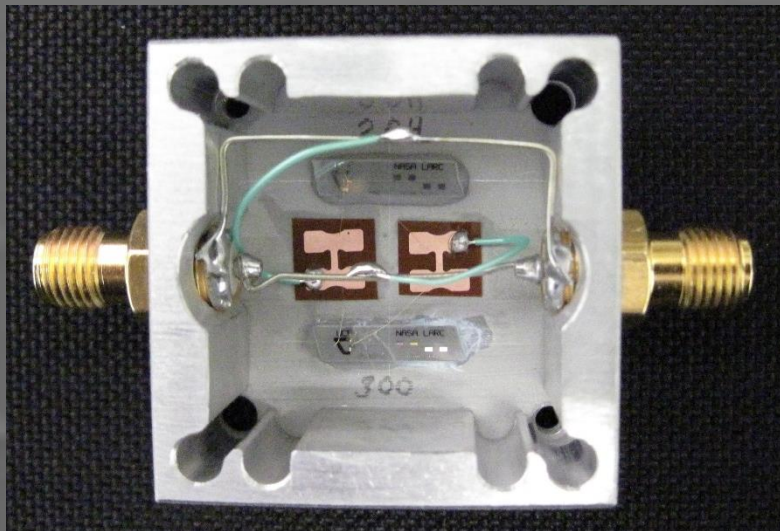
- **Increase the Number of Sensors**
  - Envision hundreds of SAW sensors mounted directly to the internal structure.
  - Reduce sensor power to enable more sensors onboard aircraft.
- **Reduce Wiring**
  - FDM allows for all sensors to be individually interrogated on a single coaxial cable.
  - Power and signals use the same coaxial cable.



# SAW Frequency Division Multiplexed (FDM) Sensor Modules



SAW sensor module  
3.81cm x 3.81cm x 1.9cm  
(1.5" x 1.5" x 0.75")



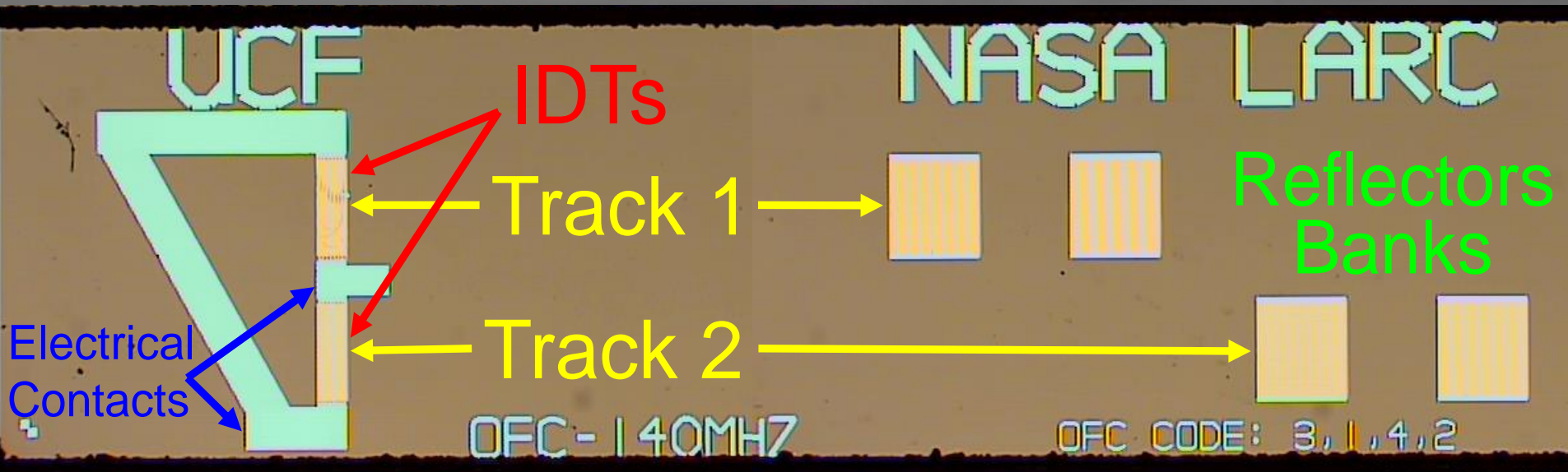
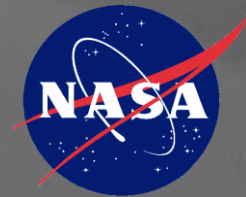
Two SAW die bonded inside, and two SMA connections.

- The prototype FDM system has six SAW sensors (140, 172, 204, 236, 268, 300 MHz) in three modules. 140/236, 172/268, 204/300.
- One SAW sensor is bonded rigidly to the bottom of the package and is used for measuring strain (236, 268, 300)
- The other SAW sensor is bonded with a flexible bonding agent such as RTV that does not transfer strain (140, 172, 204), for measuring environmental effects.





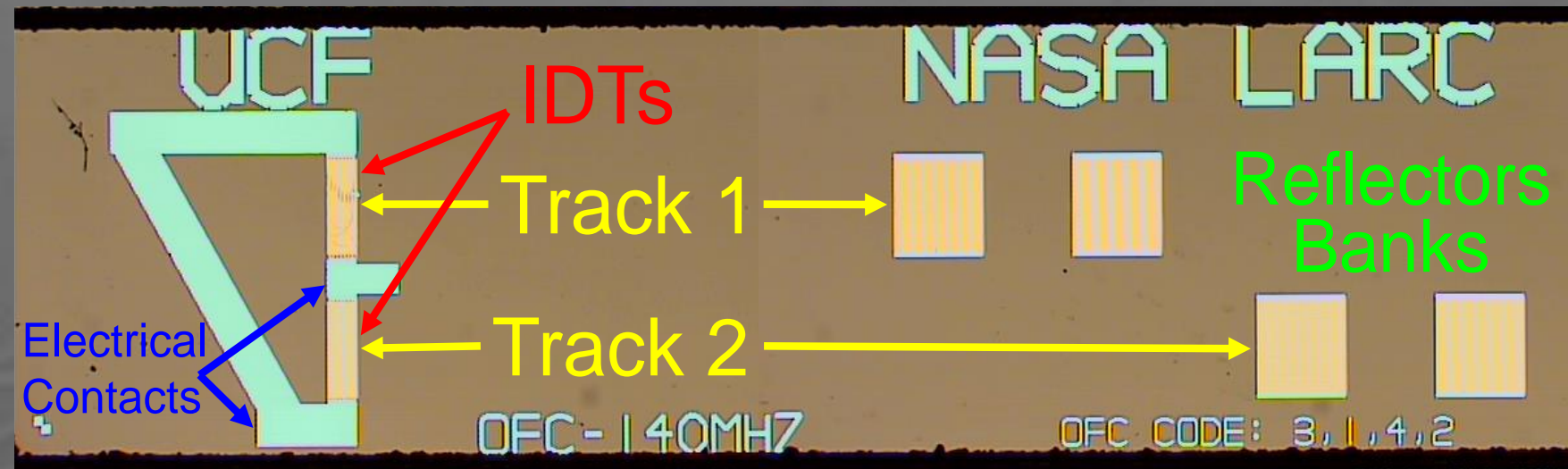
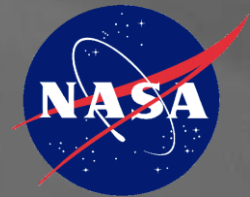
# Surface Acoustic Wave Sensor



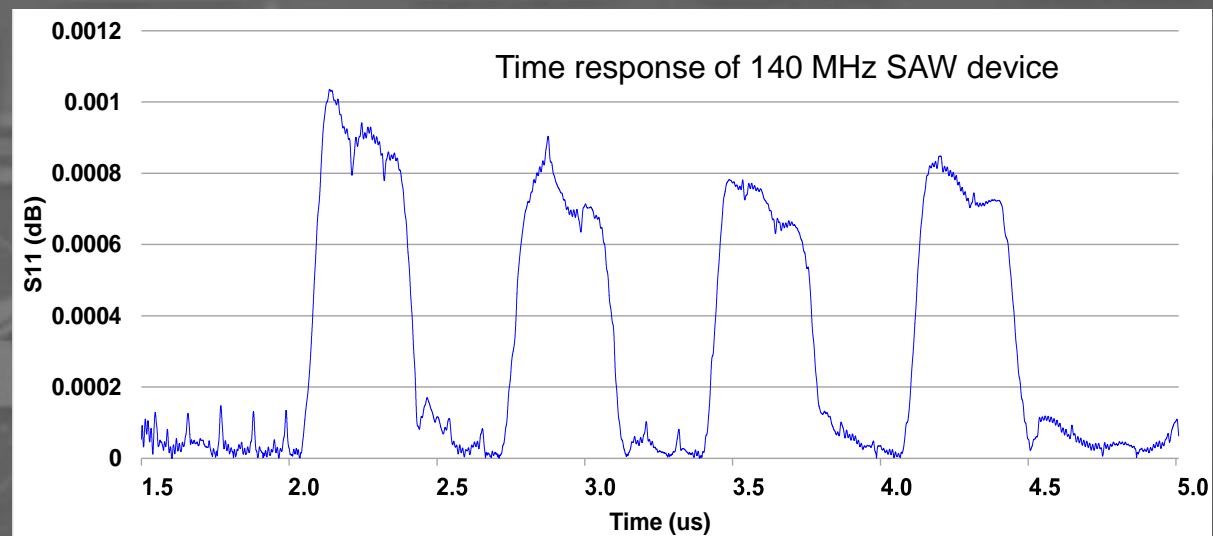
- The sensor has four orthogonal frequency coded (OFC) reflector banks in two tracks,
- The IDTs have a wider bandwidth of 15 MHz, while each reflector has a bandwidth of 6MHz.
- The gratings in each track reflect a different frequency with an arrangement of  $f_3, f_1, f_4, f_2$ .
- The reflector center frequencies are  $f_1=135.5$ ,  $f_2=138.5$ ,  $f_3=141.5$ ,  $f_4=144.5$  MHz



# Surface Acoustic Wave Sensor

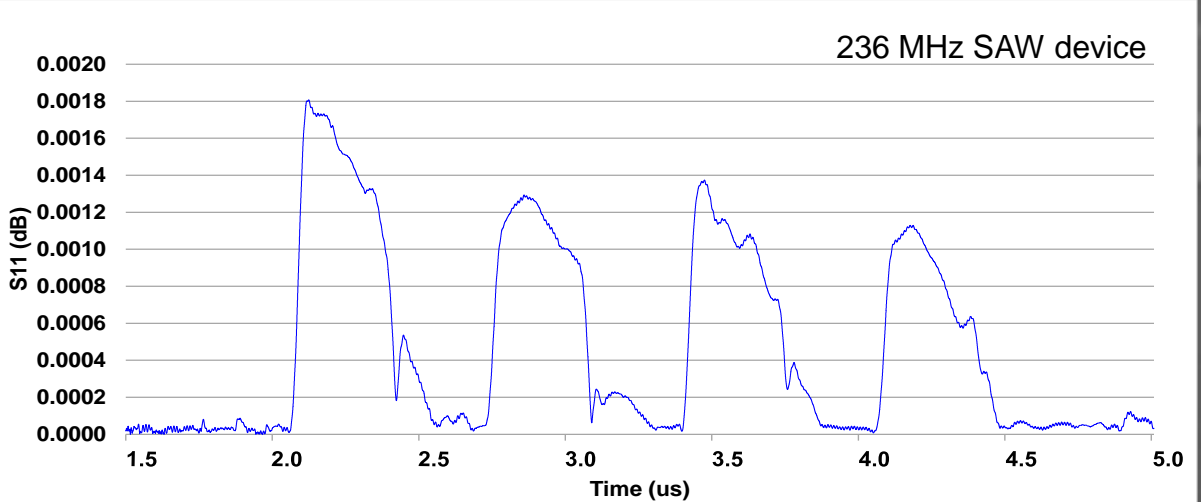
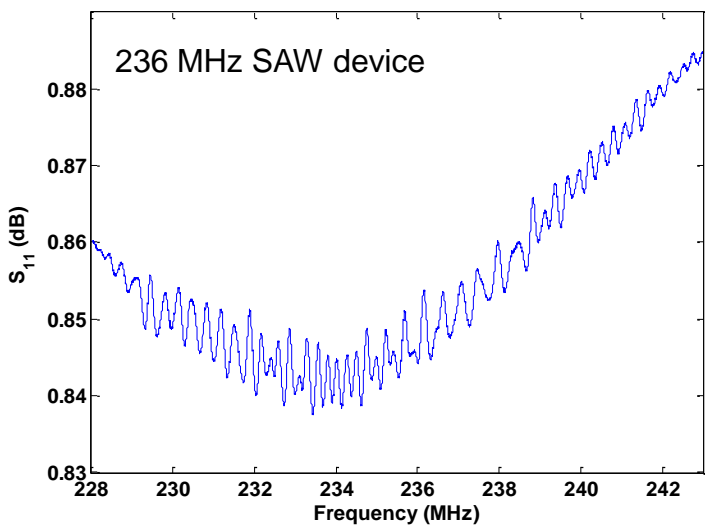
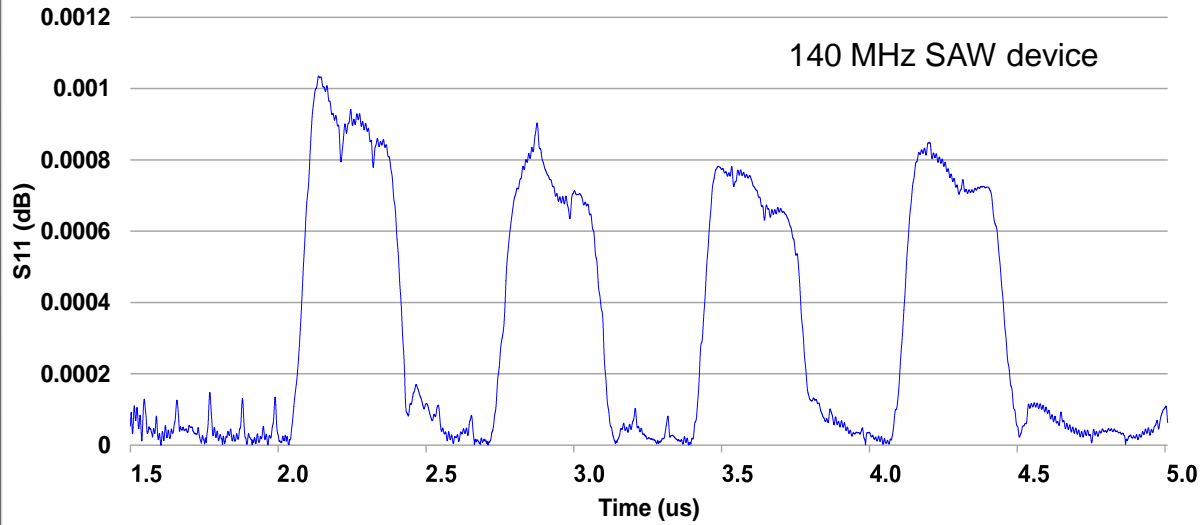
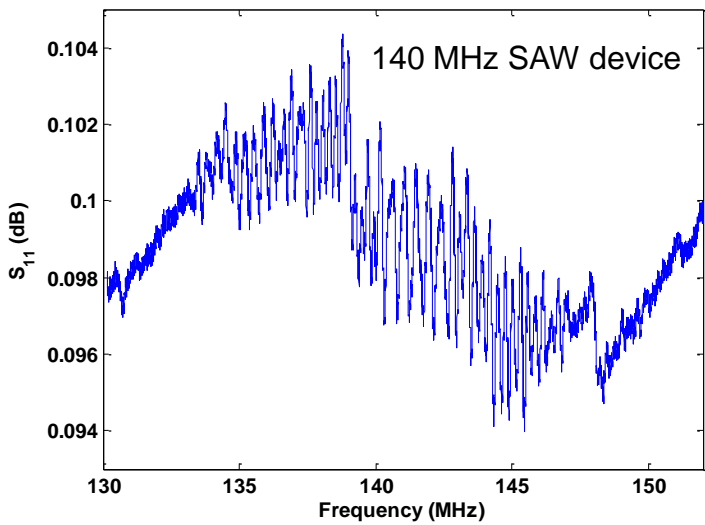
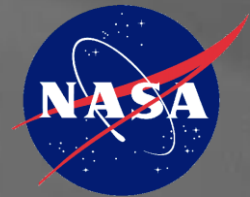


- Time Response for the four gratings.





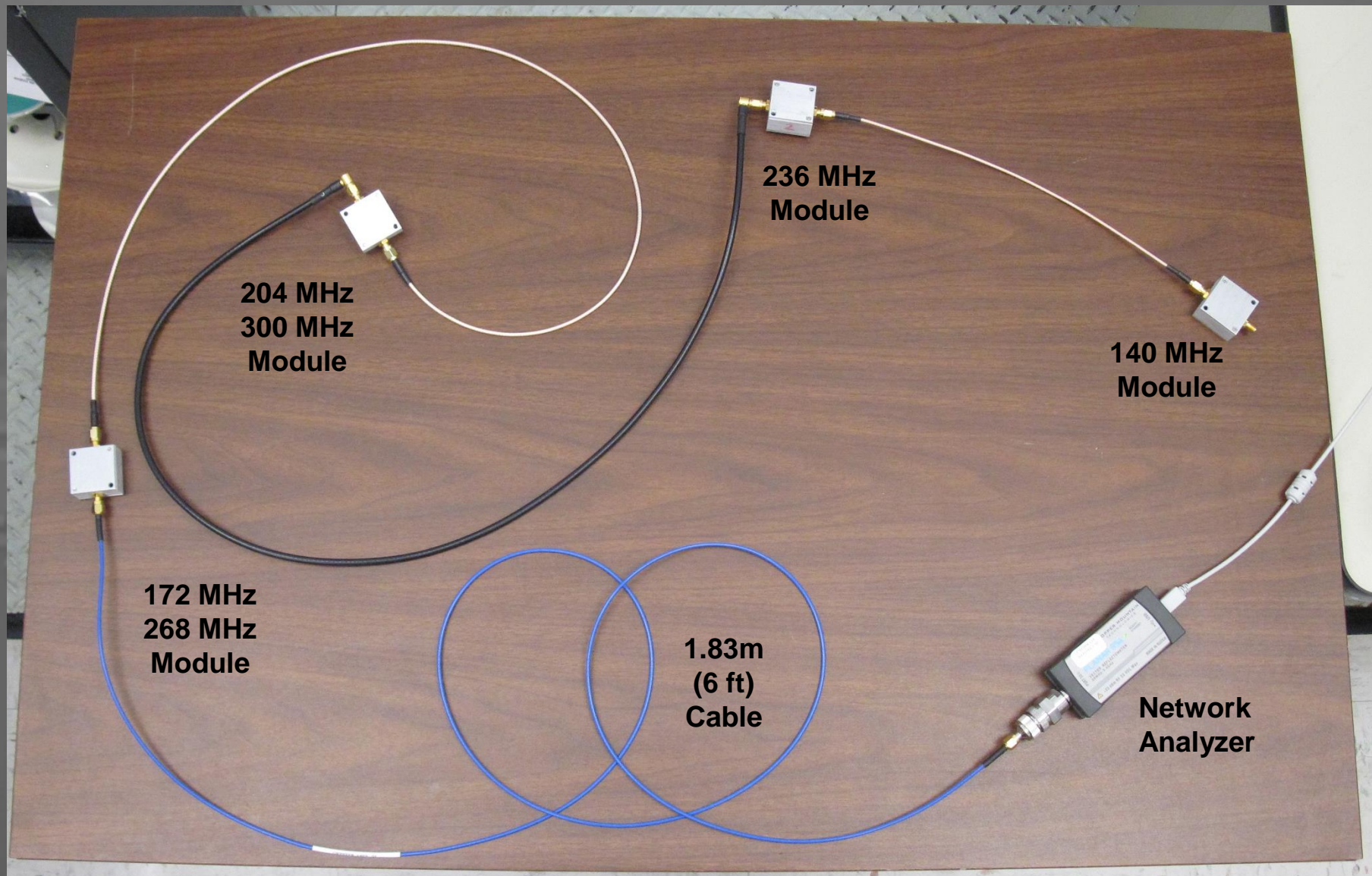
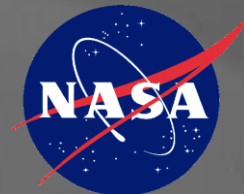
# $S_{11}$ Frequency and Time Response





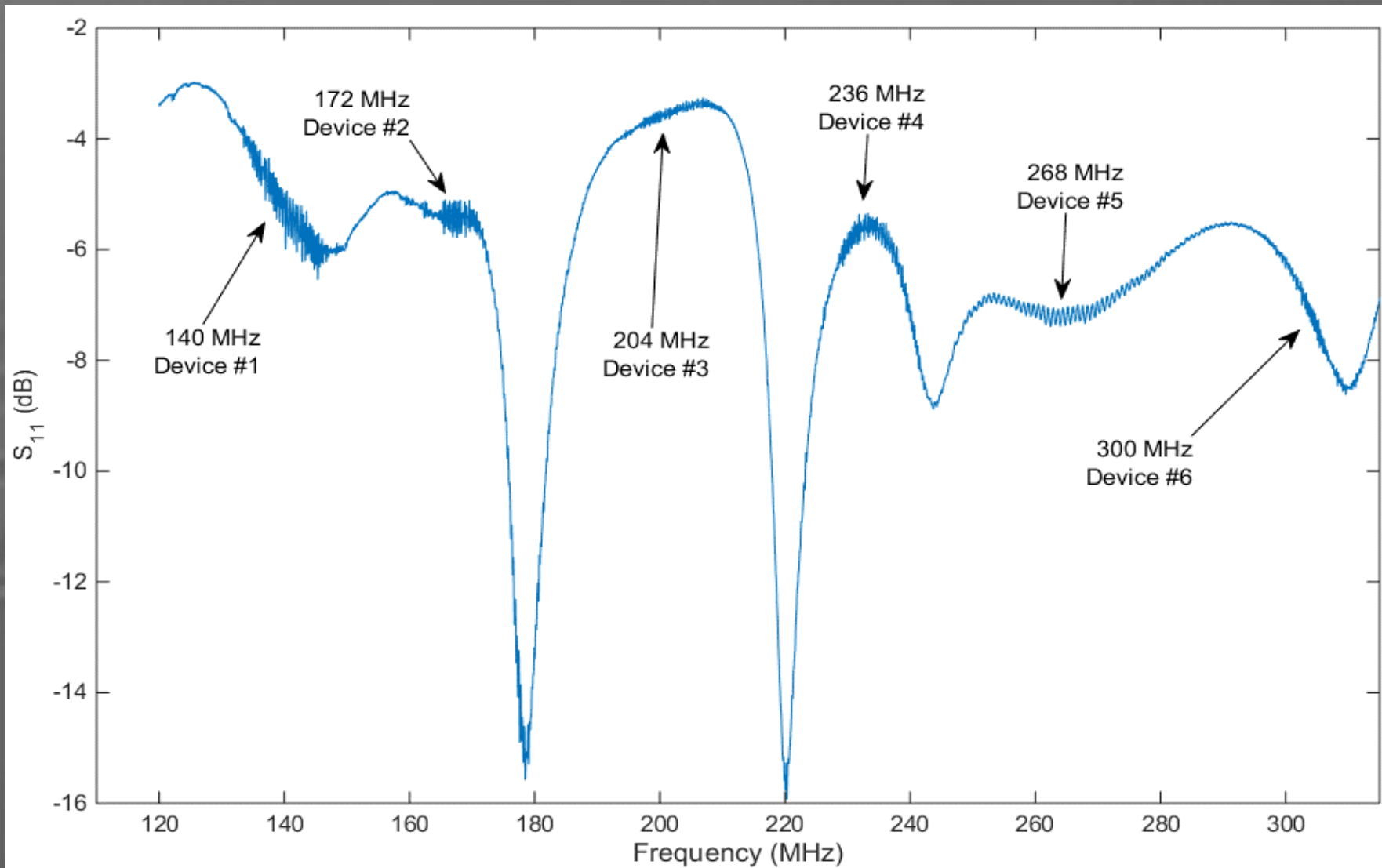
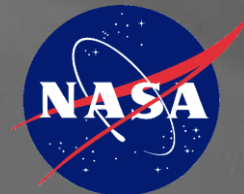


# FDM SAW Sensor System



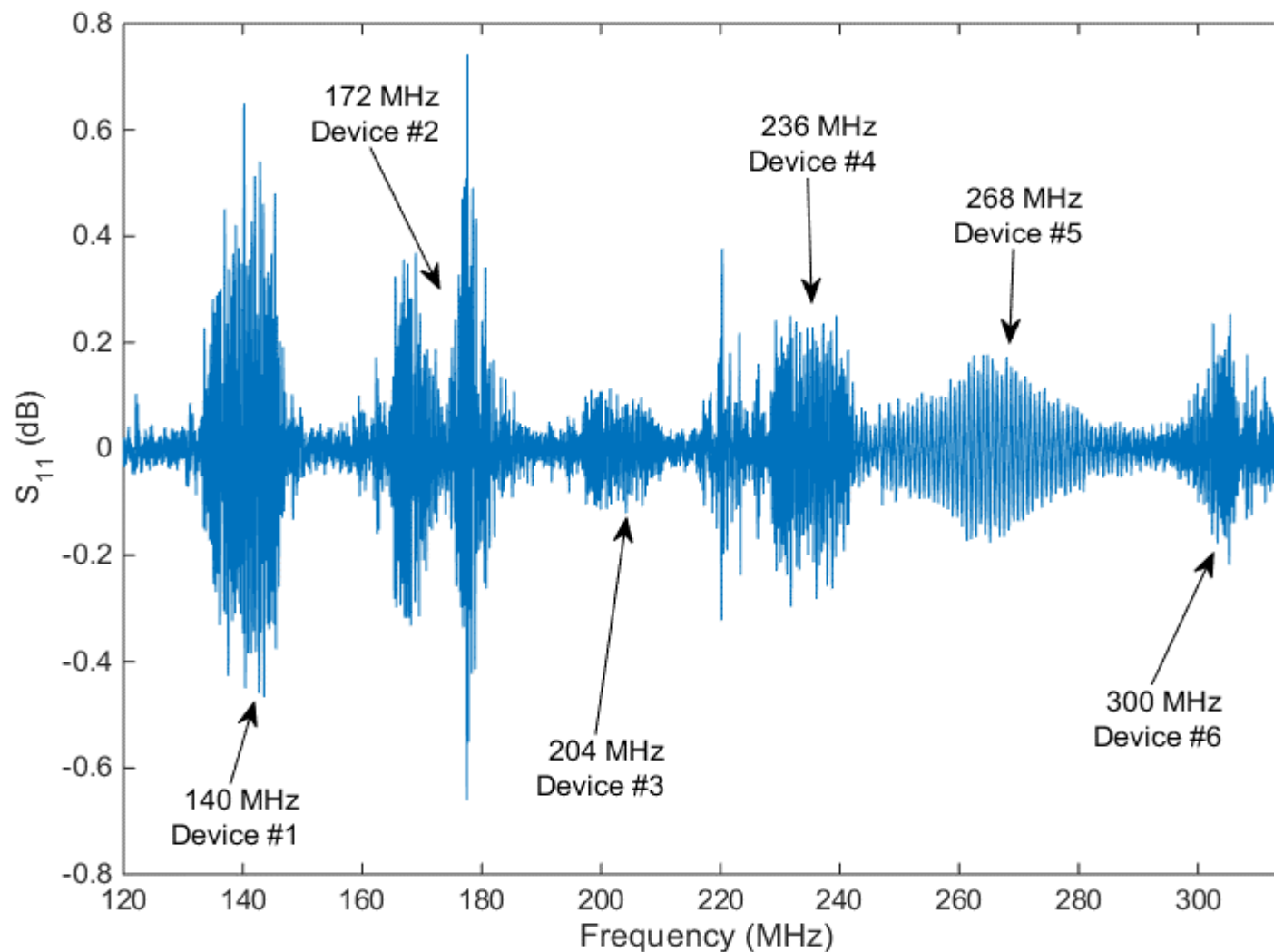
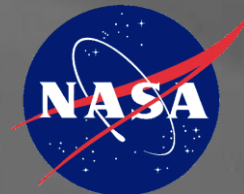


# Frequency Division Multiplexed $S_{11}$ Response



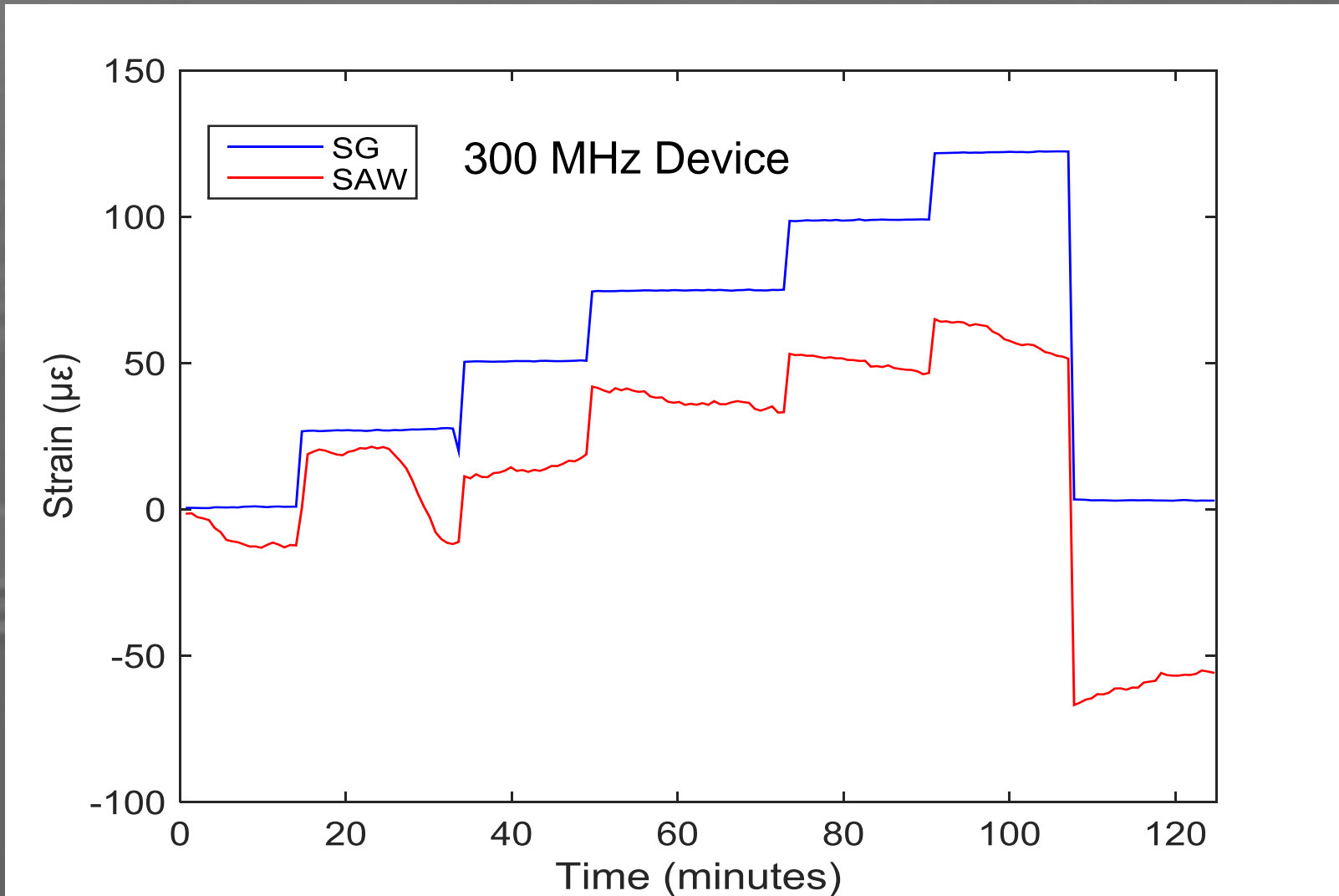
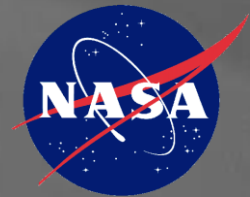


# Frequency Division Multiplexed Filtered S11 Response



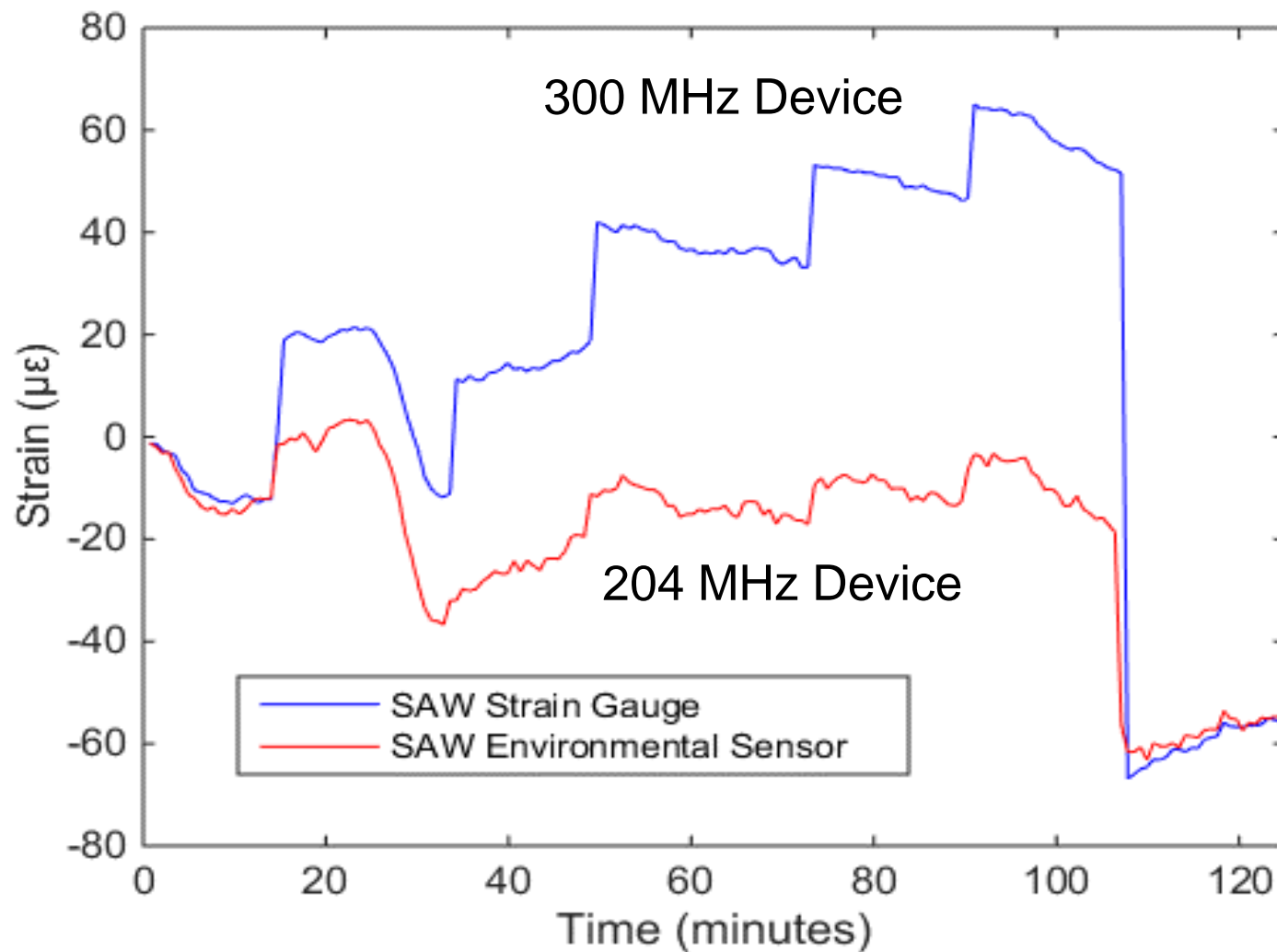
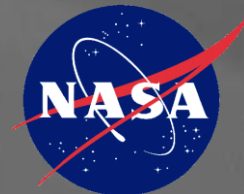


# Uncompensated Strain Results SAW vs Strain Gauge





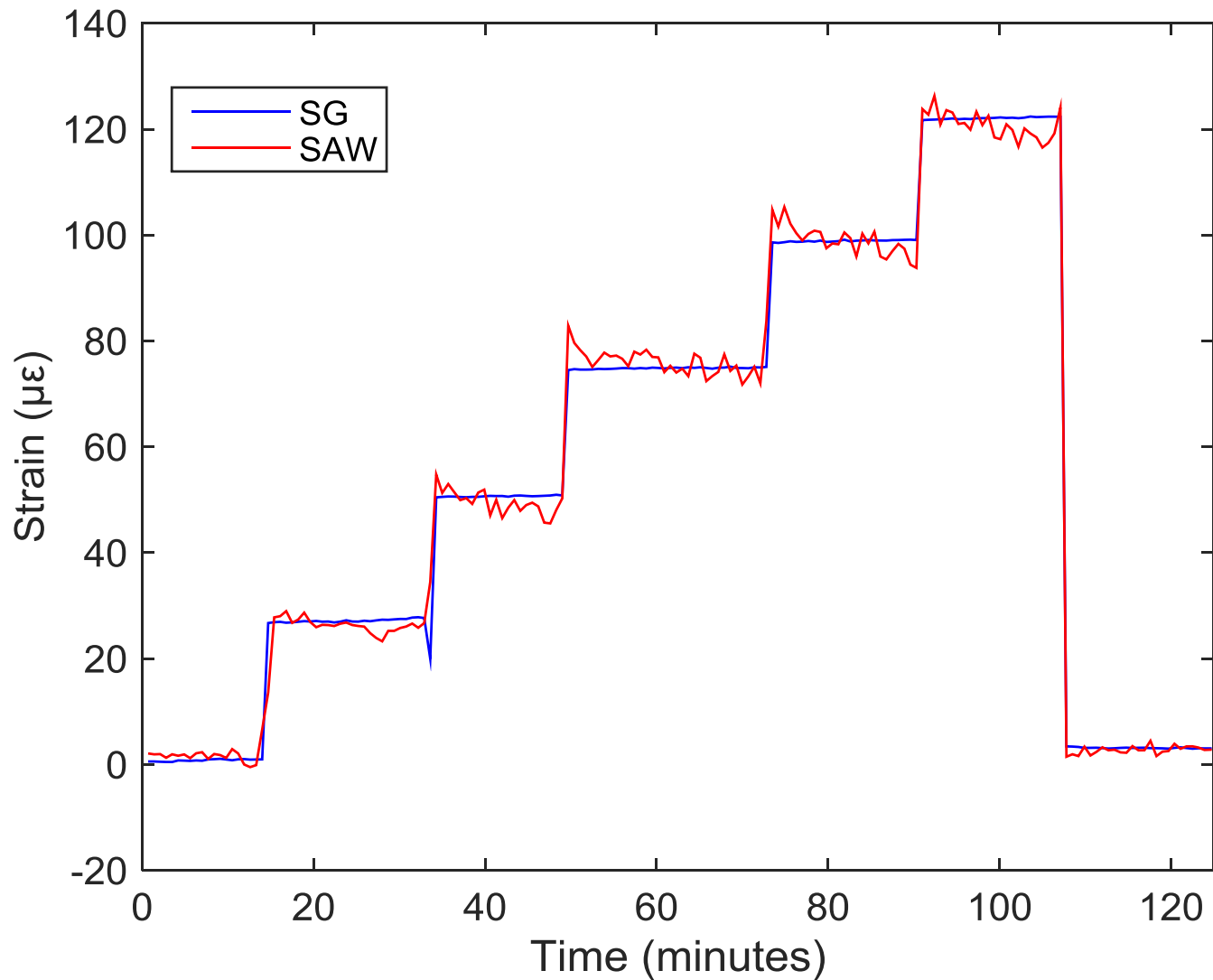
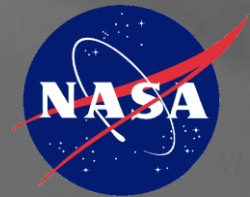
# Environmental Response SAW Sensor vs SAW Environmental





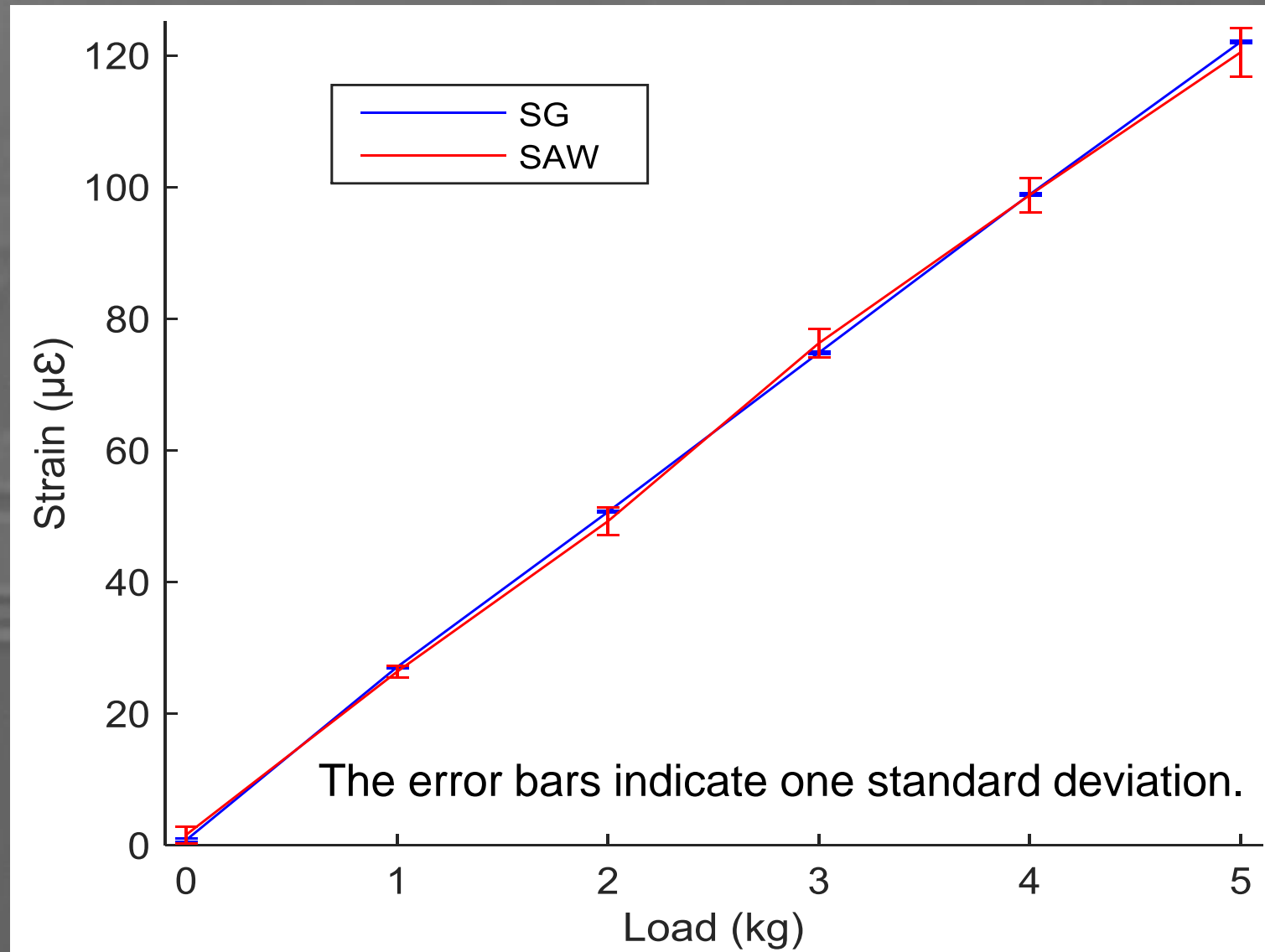
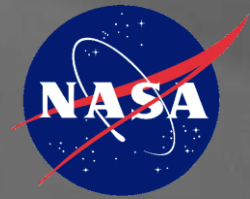


# Compensated Strain Results SAW vs Strain Gauge



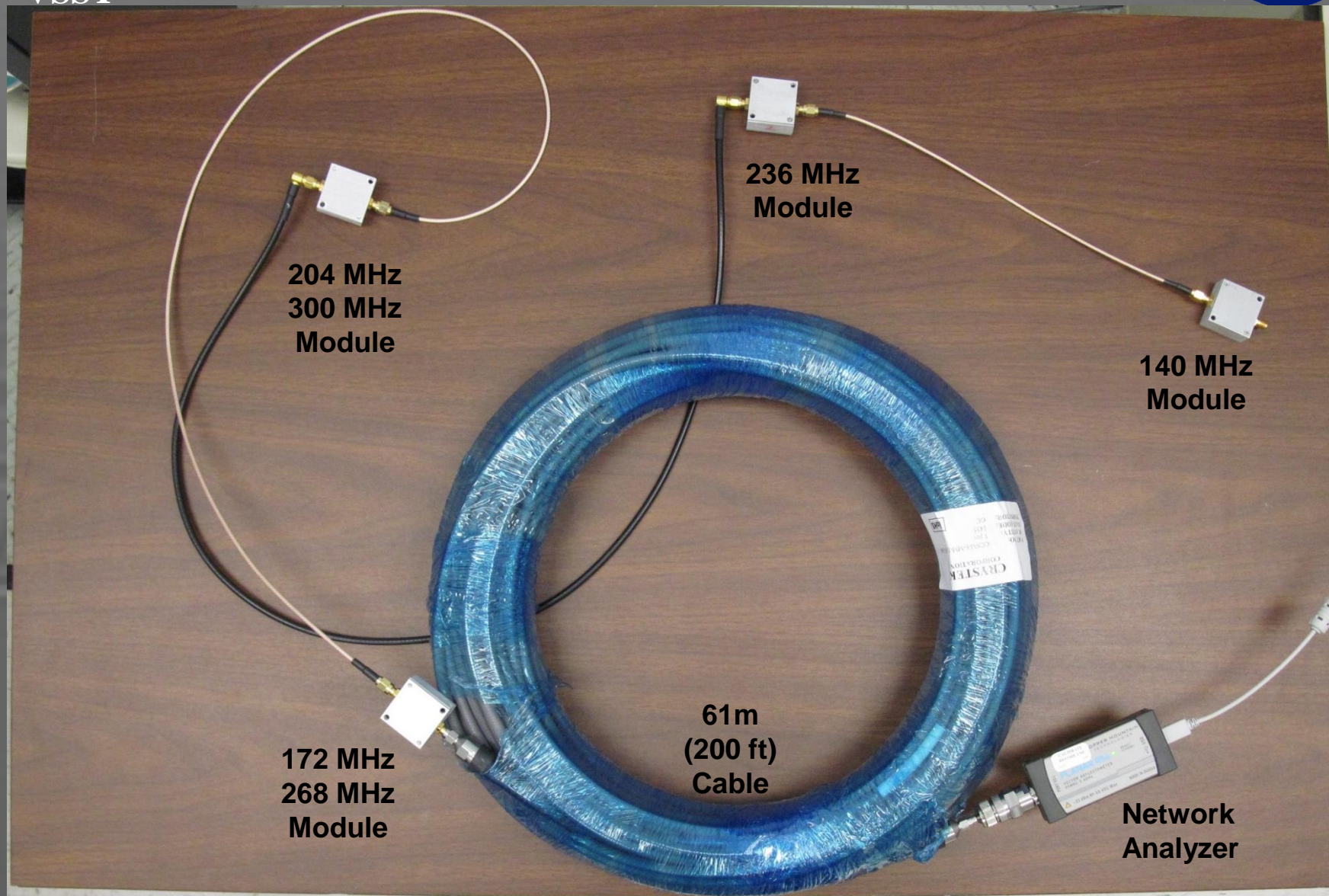
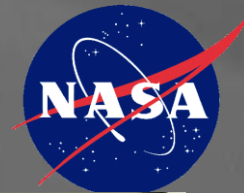


# Average Compensated Response SAW vs SG



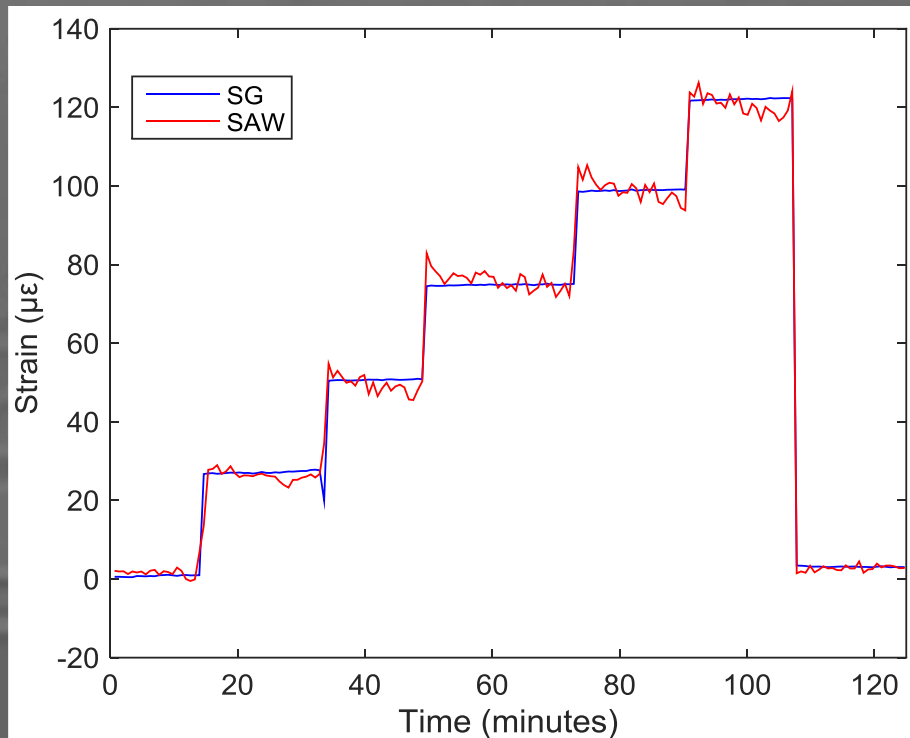
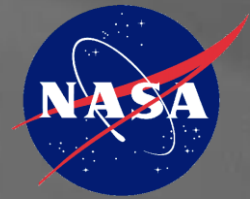


# FDM SAW Sensor System

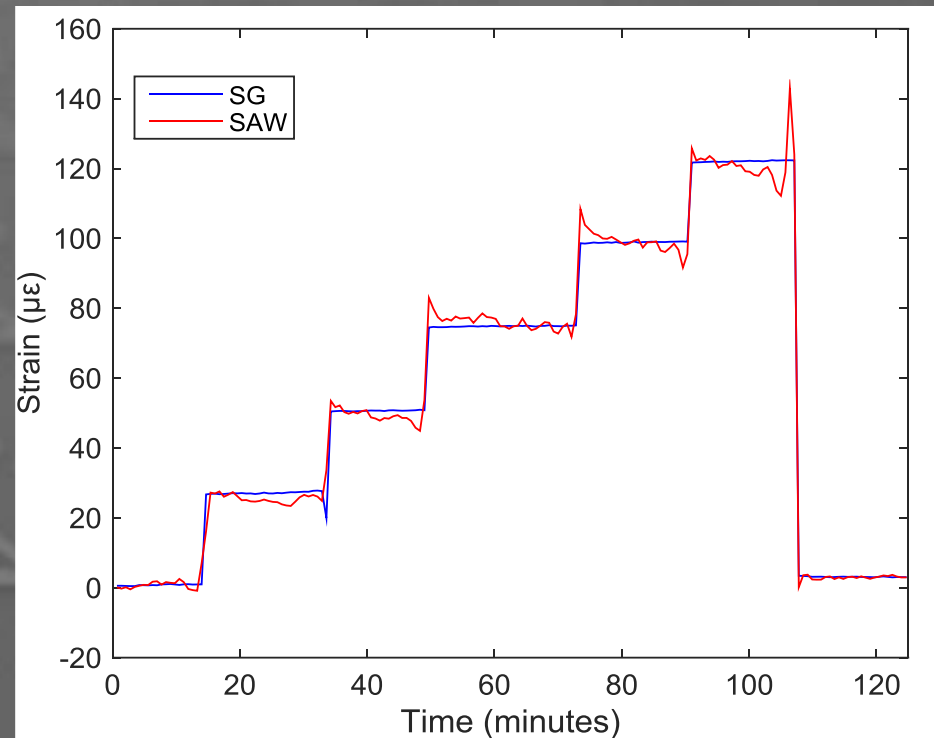




# Compensated Strain Results SAW vs Strain Gauge



**Short Cable Response**  
**1.83m (6 foot)**



**Long Cable Response**  
**61m (200 foot)**



# Future Work

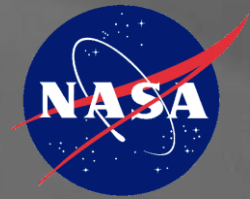


- Need optimized wire bonding techniques that will not crack the crystal and make stronger connections.
- Need impedance matching ( $50\ \Omega$ ) of the modules for more consistent behavior.
- Need to investigate ways to increase the signal to noise response of the SAW device.
- New testbed with multiple sensors (SG, TC, SAW).





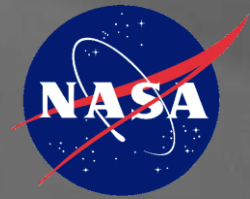
# Conclusions



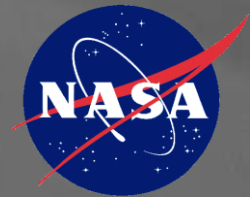
- **Demonstrated operation of 6 FDM SAW sensors, in four modules.**
- **Demonstrated wiring reduction by using a single coaxial cable for all power and signals on the four modules.**
- **Demonstrated compensated strain measurements**
  - Using short 1.83m (6ft) cable.
  - Using long 61m (200ft) cable.



# Funding/Partnership Opportunities



- **NASA does not have the resources to develop all of the sensors it needs for its applications, therefore, we are looking for partners!**
- **Small Business Innovation Research (SBIR) Small Business Technology Transfer (STTR)**
  - <http://sbir.gsfc.nasa.gov/>
  - **SBIR H13 Non-Destructive Evaluation**
    - H13.02 NDE Sensors
  - **STTR T12 Materials, Structures, Mechanical Sys. and Man.**
    - T12.01 Advanced Structural Health Monitoring
- **NASA Research Opportunities (NRAs) Grants & Contracts**
  - <http://nspires.nasaprs.com/>
- **Space Act Agreements (SAA)**
  - Partnerships with and without exchange of funds

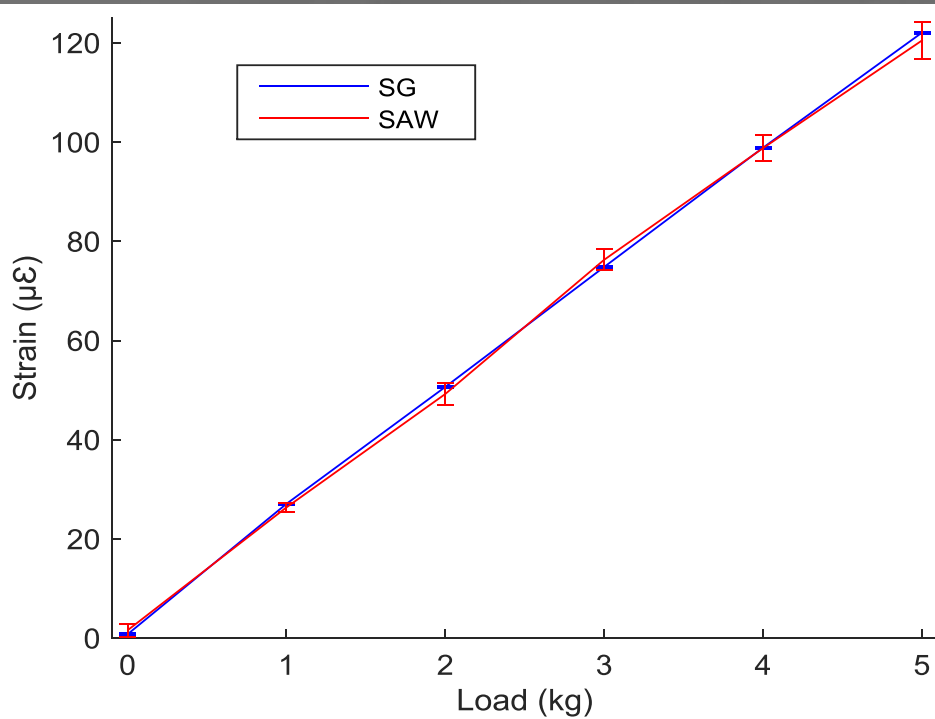
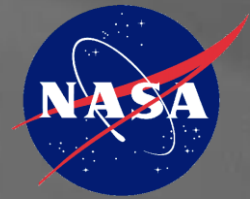


# Auxiliary Slides

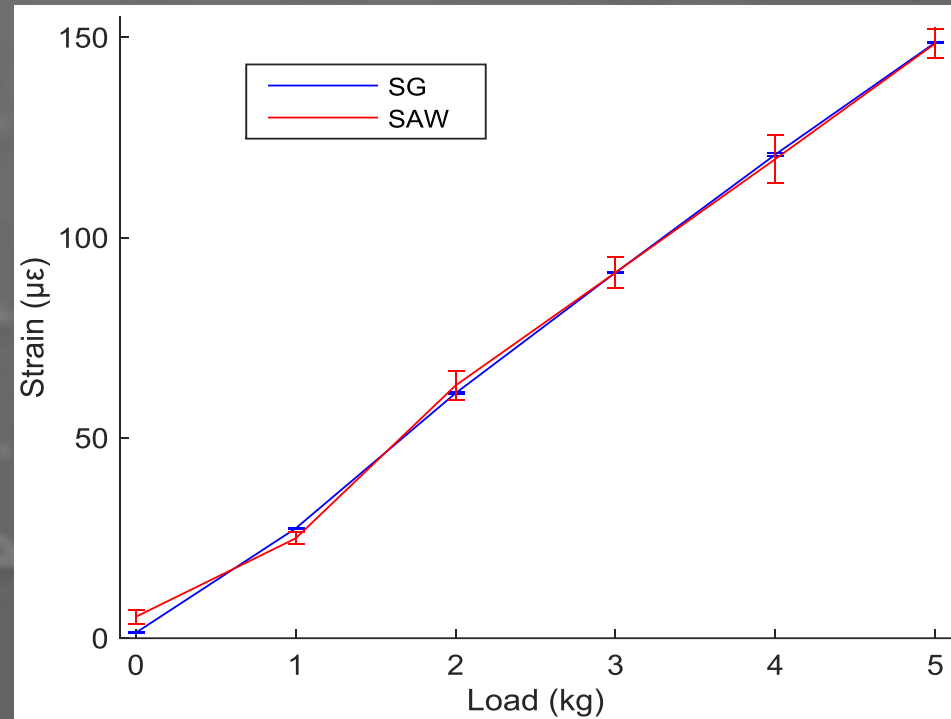




# Distance Response SAW vs SG

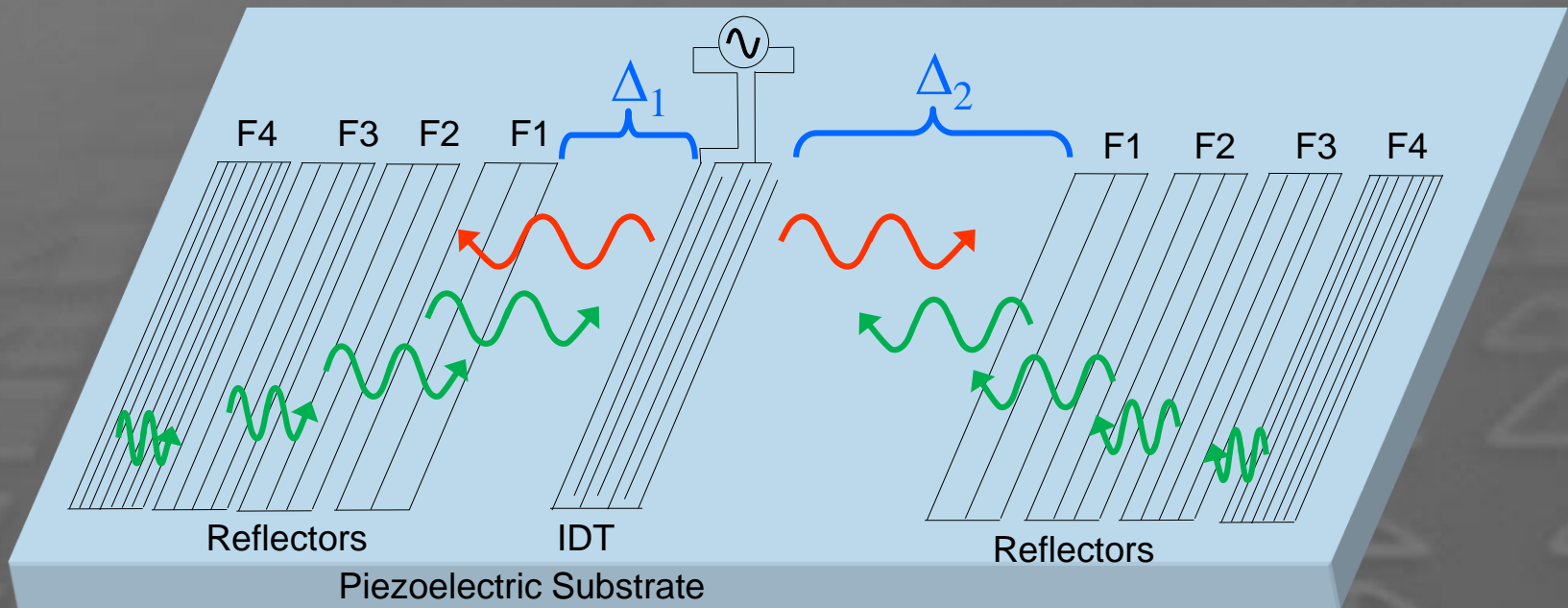


**Short Cable Response**  
**1.83m (6 foot)**



**Long Cable Response**  
**61m (200 foot)**

# Single IDT OFC SAW Sensor

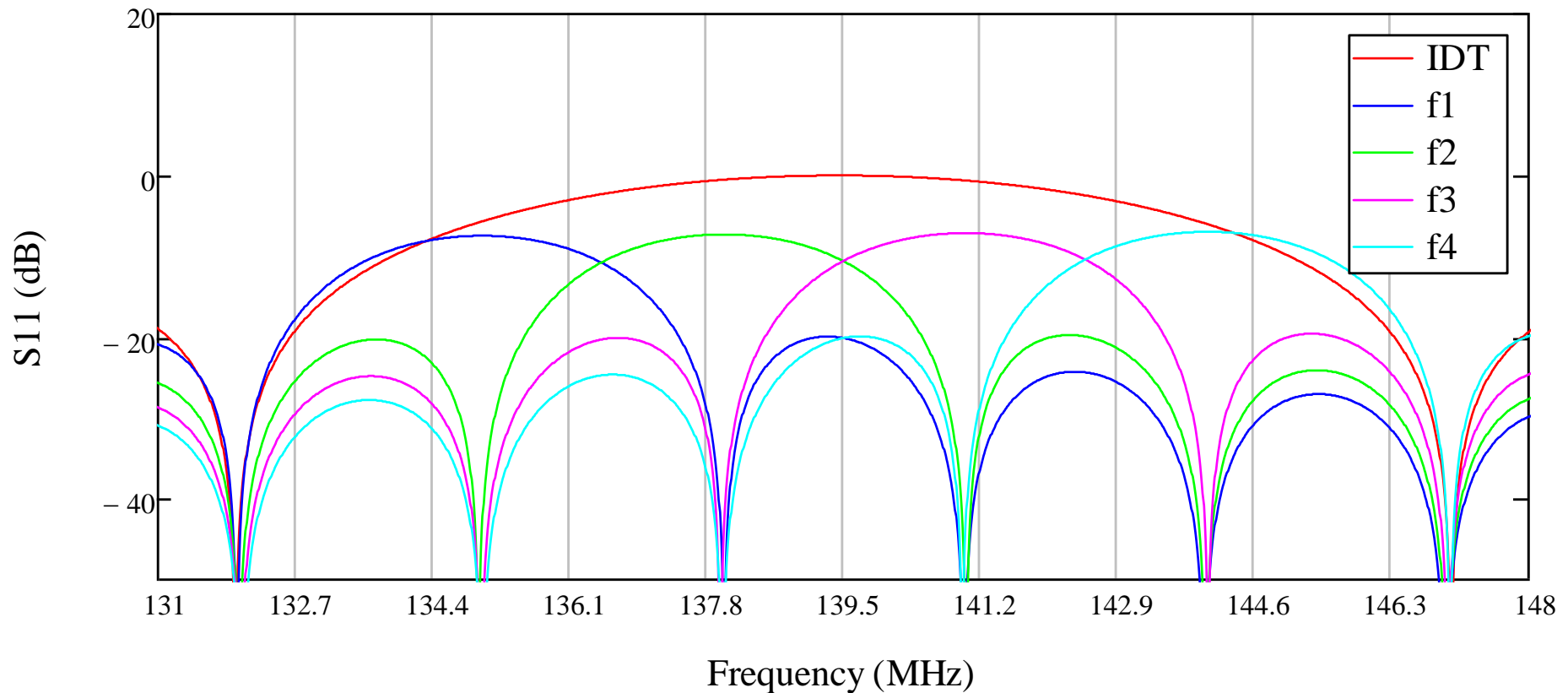
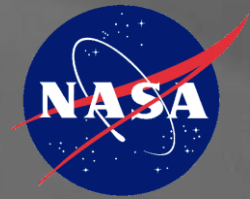


- SAW sensor that employs four orthogonal frequency coded (OFC) reflectors in two banks.
- Broadband signal generates SAW waves from the IDT (red arrows).
- Each reflector grating reflects a single frequency back (green arrows).
- $\Delta_1$  and  $\Delta_2$  are the spacings between the reflector banks and the IDT.
- $\Delta_2 > 2\Delta_1$  so ensure the reflector banks responses do not overlap in time.
- The reflected signals change frequency in response to physical changes.
- OCF uniquely codes each sensor and is Spread Spectrum (multiple frequencies).





# 140 MHz OFC Device



**The peak of each reflector occurs at the nulls of the others.  
The IDT bandwidth must cover all of the reflectors.**

# Module Wiring

